

Strengthening of Bridge Components Using Epoxy-bonded Fiber Reinforced Polymers (FRP) and Health Monitoring using Non-Destructive Evaluation (NDE) Techniques

By

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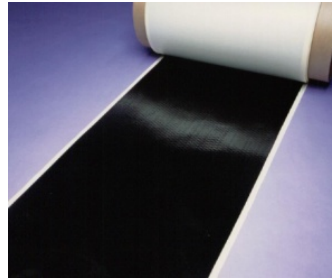
California Department of Transportation, Office of Bridge Design

Western Bridge Engineer's Seminar, Sept. 2009

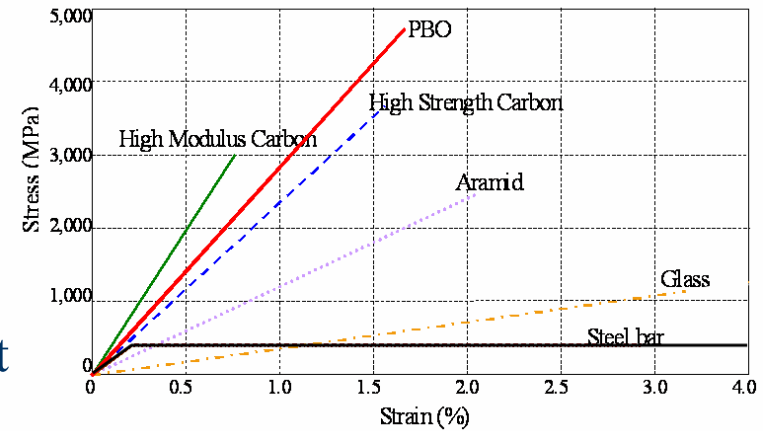
Part 1-Introduction



Fibers



Carbon Fiber Sheet



- Externally bonded **Fiber Reinforced Polymer (FRP) composites** used for strengthening of deficient RC structural components

Advantages in using composites for structural rehabilitation

- Reduced weight
- High stiffness/weight, strength/weight ratio
- Corrosion resistance
- Significant enhancements in strength and ductility
- Limited disruption of traffic

Simply Supported Beam; 35% Upgrade in Live Load

Bonded Steel Plate	Overlay Jacking	FRP Sheet
◆ 4.8mm bolted plate	◆ 2Φ8 rebar, 101.8mm grout	◆ 1 layer resin bonded
◆ 111.1kg dead load	◆ 1.1ton dead load	◆ 2.7kg dead load
◆ Placed by lift truck	◆ Formed and cured	◆ Placed by hand

Part1- Field applications



Bridge pier- confinement



Girder Strengthening



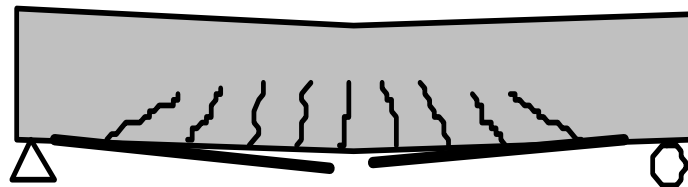
Tunnel lining



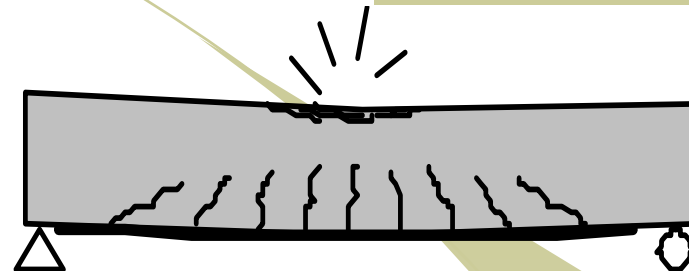
Deck Strengthening



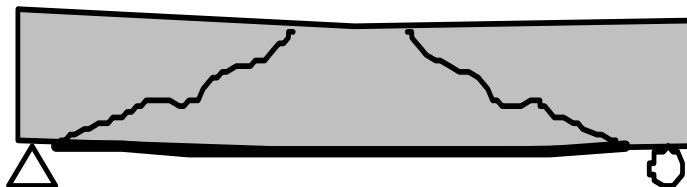
Part1-Typical failure modes



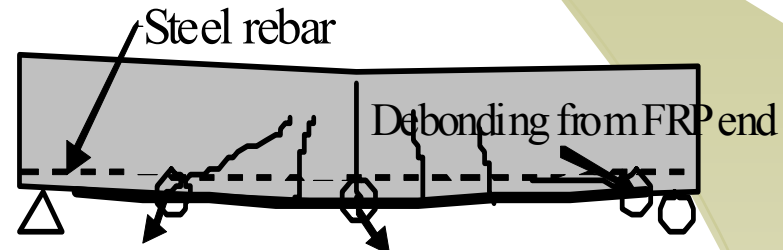
(a) FRP Rupture



(b) Concrete Crushing



(c) Concrete Shear Failure



Debonding from shear crack **Debonding from flexural crack**

(d) Bond Failures

FRP Flexural Strengthening of Structures

Intermediate crack-induced debonding failure
is dominant and hard to predict

Part 1- ACI 440 procedure 1

- ACI nominal capacity is limited by the effective FRP strain, governed by concrete crushing ($\epsilon_{cu} = 0.003$) and debonding failure

Strain in FRP at concrete crushing

Initial strain in FRP due to dead load

Ultimate strain of FRP factored by bond coefficient to prevent debonding failure

$$\epsilon_{fe} = \epsilon_{cu} \left(\frac{h-c}{c} \right) - \epsilon_{bi} \leq \kappa_m \epsilon_{fu}$$

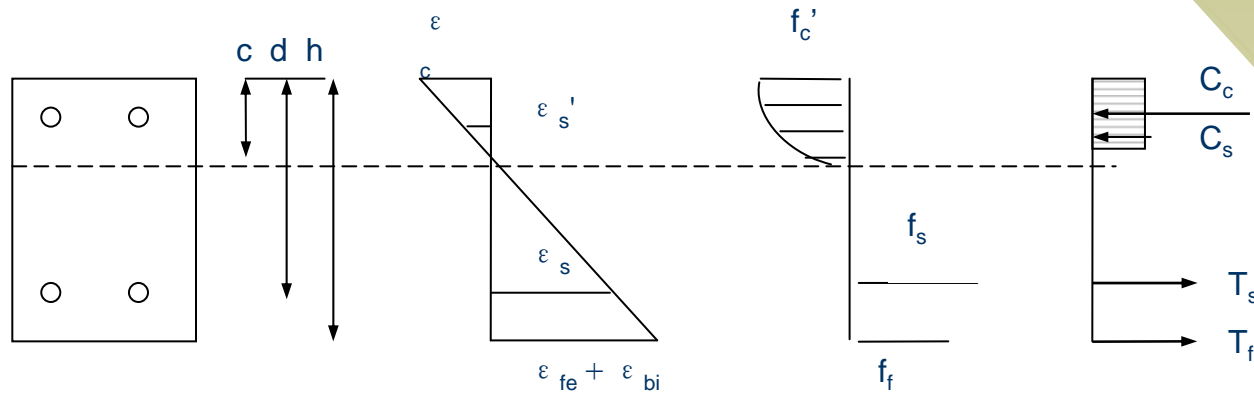
- Ultimate strain of FRP is factored by bond coefficient in order to take into account premature debonding

$$\kappa_m = \frac{1}{60\epsilon_{fu}} \left(1 - \frac{nE_f t_f}{360,000} \right) \leq 0.9 \text{ for } nE_f t_f \leq 180,000$$

$$\kappa_m = \frac{1}{60\epsilon_{fu}} \left(\frac{90,000}{nE_f t_f} \right) \leq 0.9 \text{ for } nE_f t_f > 180,000$$

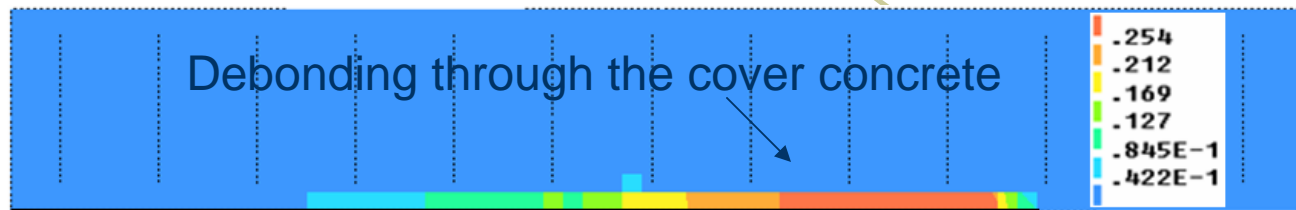
Part 1-ACI 440 procedure 2

$$M_n = A_s f_s \left(d - \frac{\beta_1 c}{2} \right) + \psi_f A_f f_{fe} \left(h - \frac{\beta_1 c}{2} \right) - A_s' f_s' \left(d' - \frac{\beta_1 c}{2} \right)$$



- ◆ **Nominal moment is calculated in the same way as conventional reinforced concrete, except there is an extra term for the FRP**
- ◆ **A reduction factor, ψ , is applied to the force in the composite, reducing its contribution to the nominal strength of the beam, in order to account for the uncertainty involved with FRP composites**

Part 1-Debonding failure mechanisms by FEM



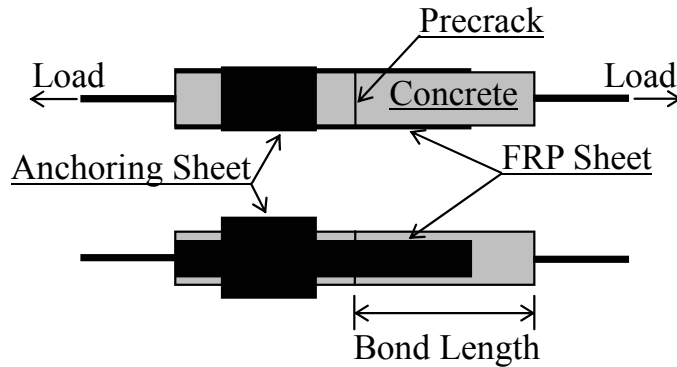
Low concrete strength



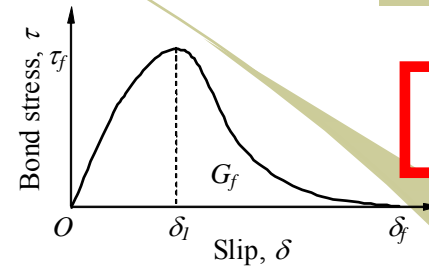
High concrete strength

Debonding is an interface-related failure associated with properties of concrete, bond and FRP

Part 1-Comparison of Loading Conditions



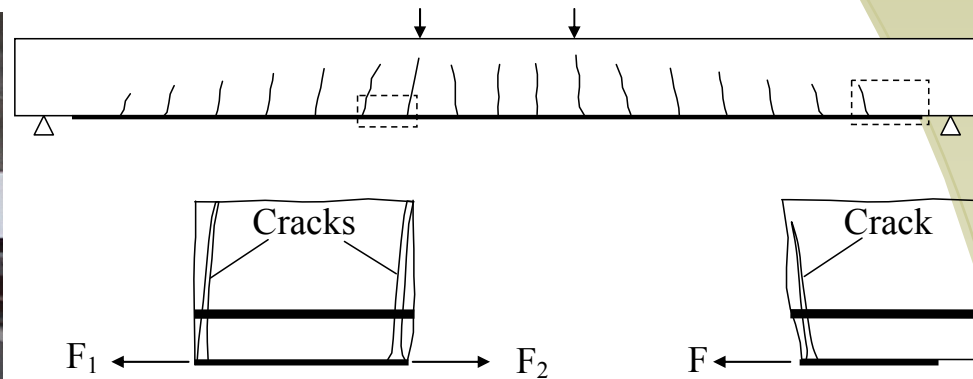
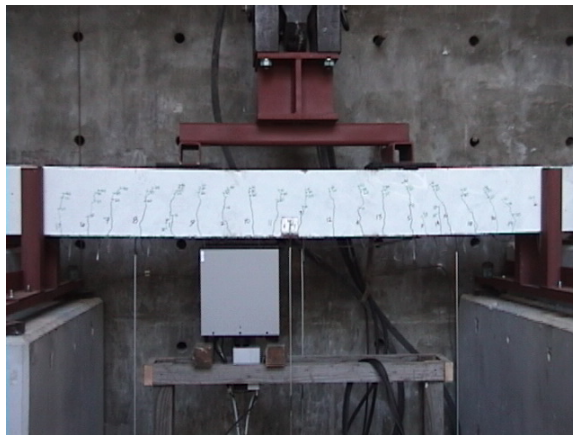
Simple bond test



$$P_{\max} = b_{frp} \sqrt{2G_f E_{frp} t_{frp}}$$

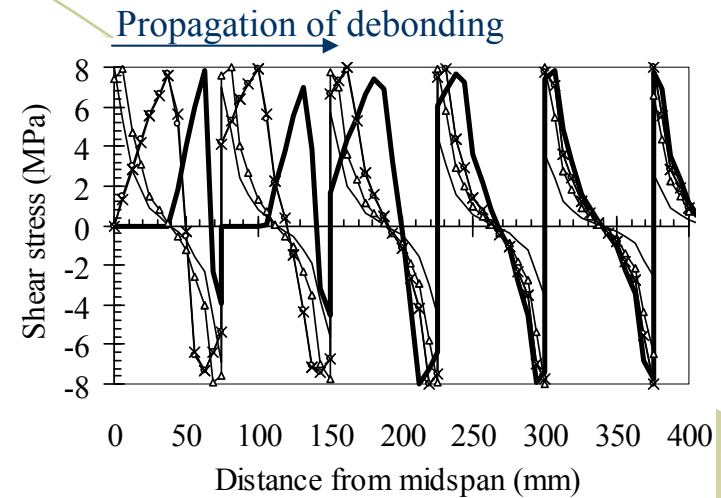
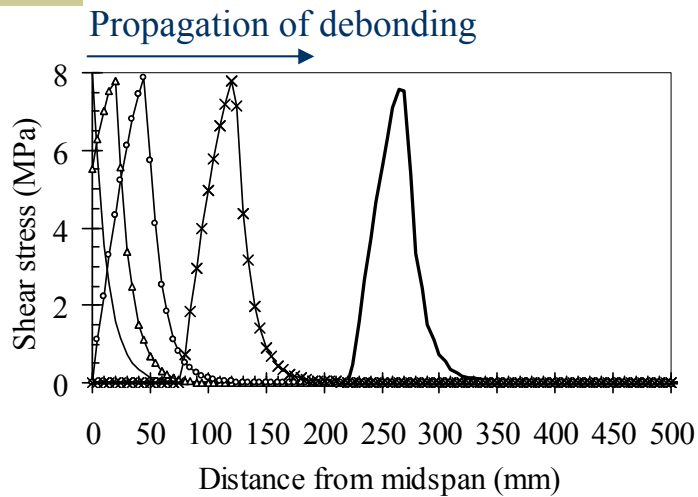
where b_{frp} , E_{frp} and t_{frp} are width, modulus and thickness of FRP, respectively;

G_f =interfacial fracture energy consumed for debonding failure.



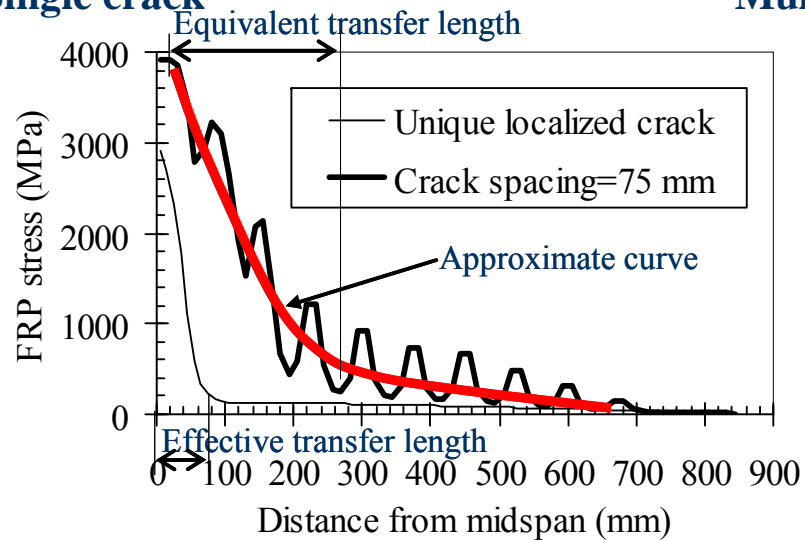
Debonding mechanisms in FRP-strengthened beams

Part 1-Debonding behaviors by FEM

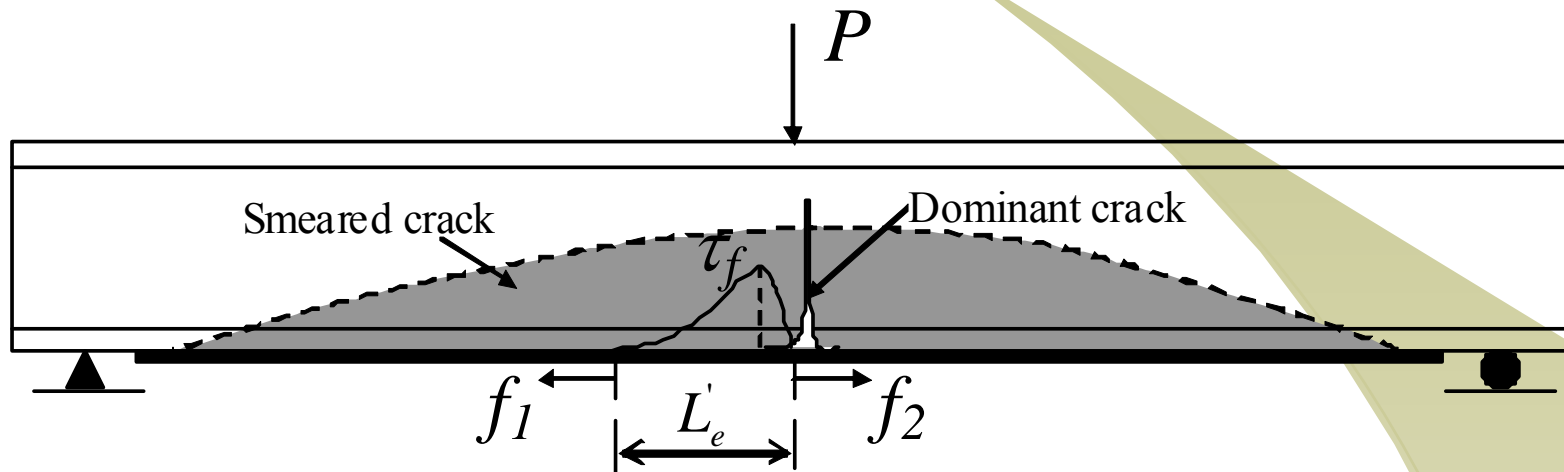


Simple bond test/ Single crack

Multiple cracks



Part 1-Prediction of debonding failure load

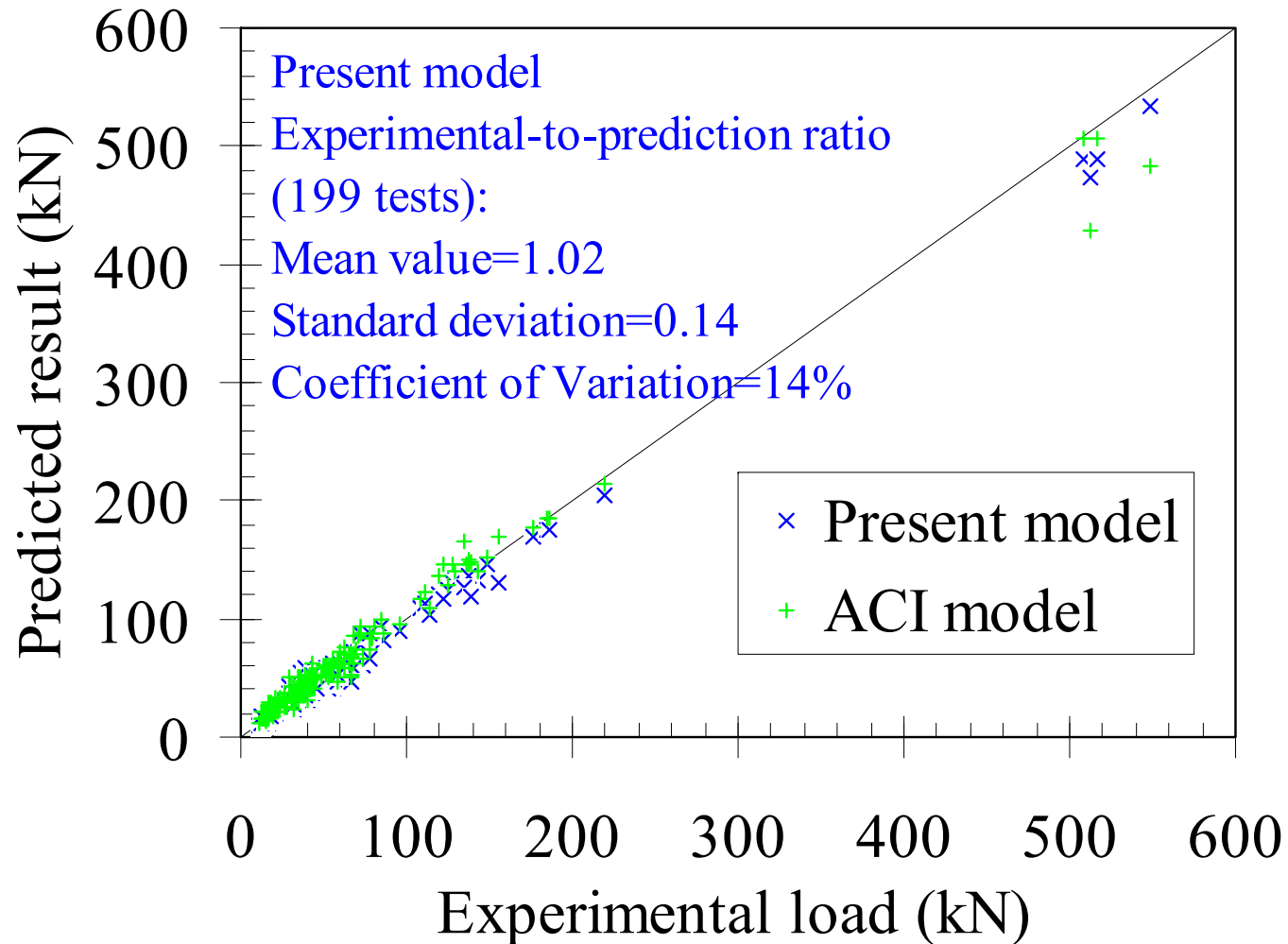


$$L'_e = \frac{1.3\sqrt{E_{frp} t_{frp}}}{f_c^{0.095}} \quad (\text{mm, MPa})$$

$$P_{deb} \Rightarrow (f_2 - f_1) \geq b_{frp} \sqrt{2G_f E_{frp} t_{frp}}$$

$$G_f = 0.644 f_c^{0.19} \quad (\text{N/mm, MPa})$$

Part 1-Validation of the proposed model



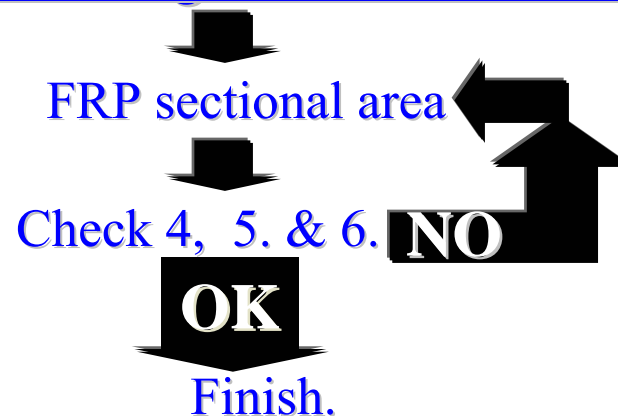
Simple model: $\varepsilon_{deb} = 0.23(f_c')^{0.2} / (E_{frp} t_{frp})^{0.35}$ (Wu et al. 2008)

Part 1-Design considerations

1. Crushing of concrete before yielding of reinforcing steel
2. Yielding of reinforcing steel followed by rupture of FRP
3. Yielding of reinforcing steel followed by concrete crushing
4. Premature failure at ends of FRP laminate
5. Debonding of FRP due to flexural cracking
6. Peeling-off of FRP caused by shear cracking

Ductile & Preferable

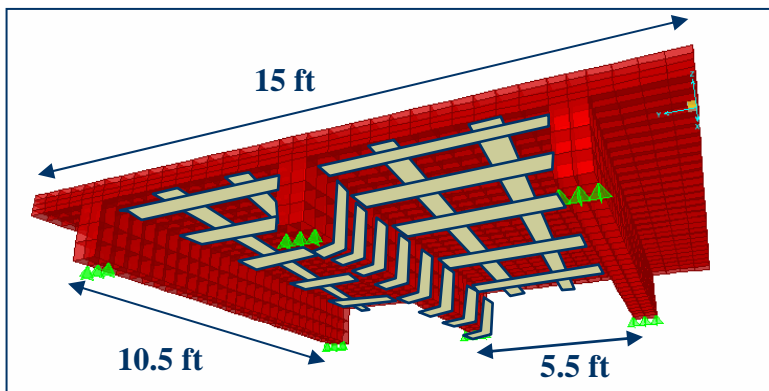
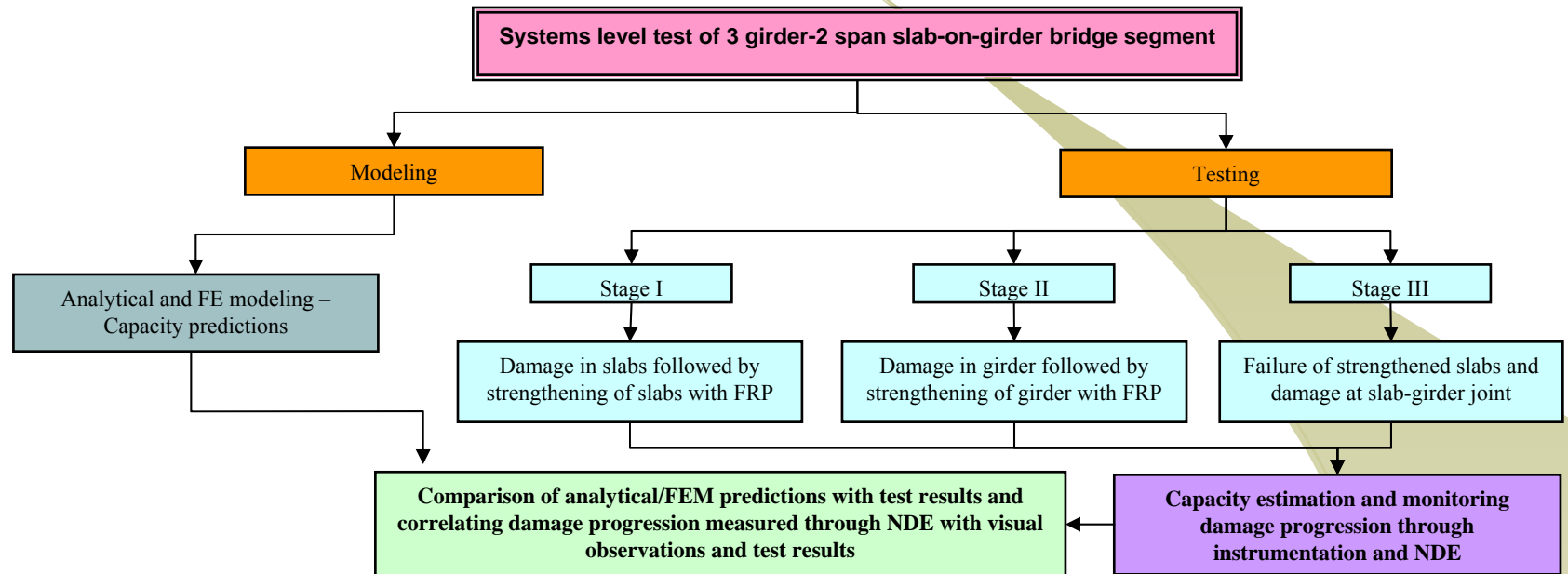
3. Yielding of reinforcing steel followed by concrete crushing



Part 2-Non-destructive evaluation

- ◆ To be used not only as a tool for periodic inspection and quality assessment but also to quantitatively monitor over time the appearance/progression of damage
- ◆ Should be rapid, cost-efficient and reliable
- ◆ **Objective of NDE in FRP rehabilitated structures:**
 - Quality Assurance / Data inventory
 - Local NDE
 - Evaluate performance level of FRP composite strengthening through detection of subsurface damage and deterioration at the composite-concrete or composite-composite interface
 - Global NDE
 - Health monitoring of overall structural performance over its life cycle in terms of damage appearance and/or progression

Part 2-Case study overview



Part 2-Composite strengthening

STRENGTHENING OF SLAB WITH COMPOSITE- Flexural strengthening



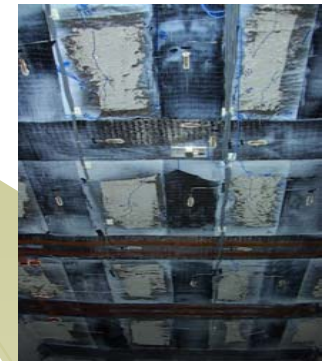
Resin impregnation of fabric



Adhesive application on concrete



Bonding fabric to concrete



Finished product

STRENGTHENING OF GIRDER WITH COMPOSITE- Shear strengthening



Application of primer coat



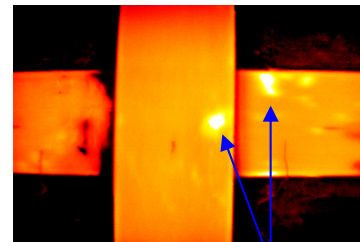
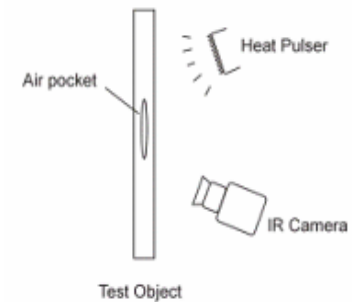
Bonding FRP to concrete



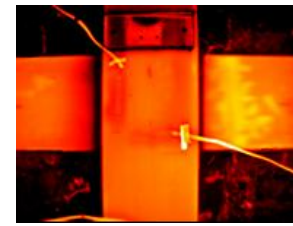
Finished product

Part 2-Local NDE – IR thermography

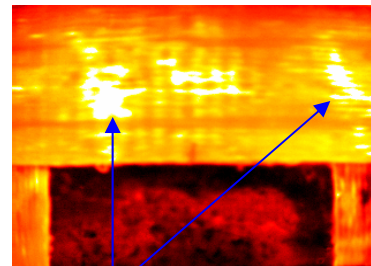
- ◆ IR thermography as NDE tool for FRP Composite Defects:
 - Non-contact optical technique aimed at detection of subsurface defects with an infrared (IR) camera under relevant temperature differentials produced through **ACTIVE** (external source) or **PASSIVE** (natural source, e.g. the sun) heating
 - Defects cause interruption in heat flow resulting in **hot / cold spots**



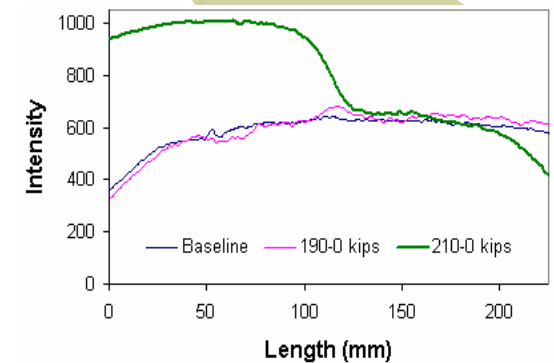
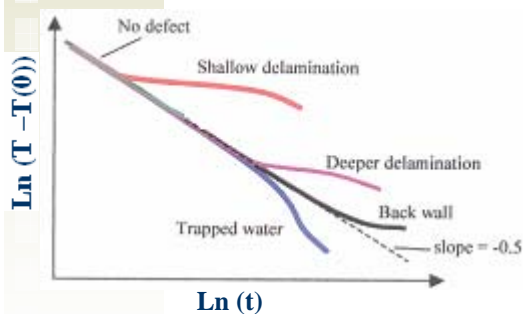
Defects at composite interface



Damage progression at a crack



Defects at composite-concrete bond



- ◆ Real-time inspection and data interpretation allow instant assessment of integrity and serviceability

Part 2-Characterization of defects

1	<p>Debonding of composite at pultruded strip - concrete interface</p>		
2	<p>Interlaminar debonding in the pultruded strip due to separation between fibers and matrix of strip</p>		
3	<p>Debond at pultruded strip -concrete interface due to opening of cracks</p>		
4	<p>Air void defects produced during installation of the pultruded strip</p>		
5	<p>No damage/defects</p>		

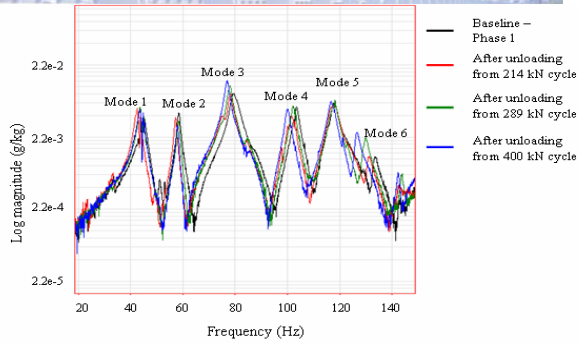
Part 2-Global NDE – modal testing

- ◆ **MODAL TESTING**
- ◆ Obtain the dynamic signature of the structure (frequencies and mode shapes)
 - Related to mass and stiffness

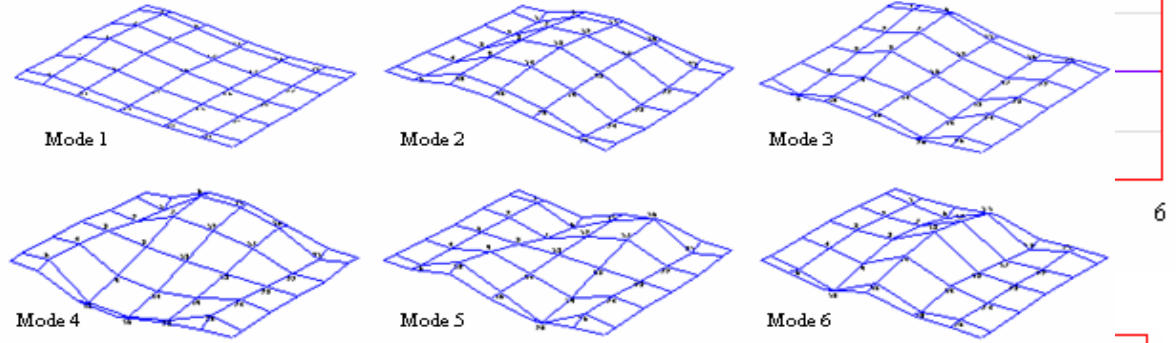
- ◆ **HEALTH MONITORING / MODEL UPDATING**
- ◆ Monitor the health (in terms of stiffness) of structure by monitoring the frequencies and mode shapes
- ◆ Calibrate a baseline finite element model based on dynamic characteristics obtained from the baseline modal tests
- ◆ Use model updating over time to **quantitatively** determine the changes in stiffness parameters in **localized regions of the model** corresponding to the changes in the frequencies and mode shapes
- ◆ Localize the effects of damage progression and strengthening through quantification of the parameter changes

Part 2-Modal testing

Data Analysis

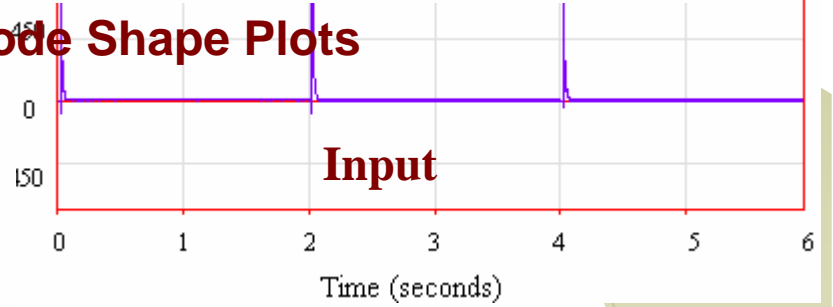
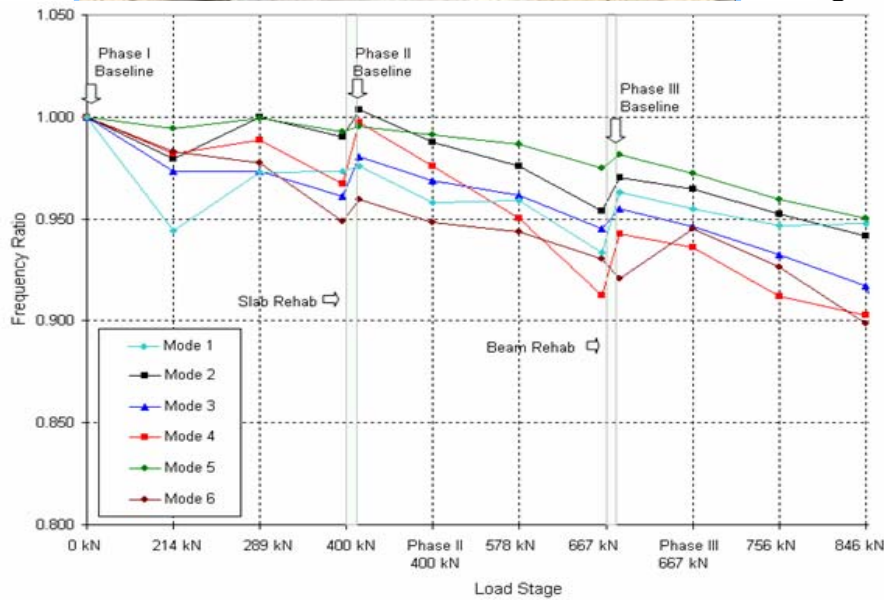


Spectral Density Plot



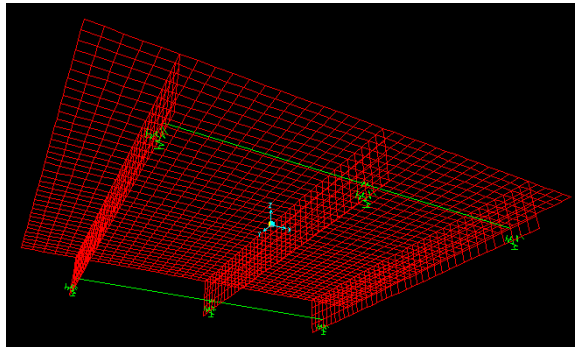
Mode Shape Plots

Visual In

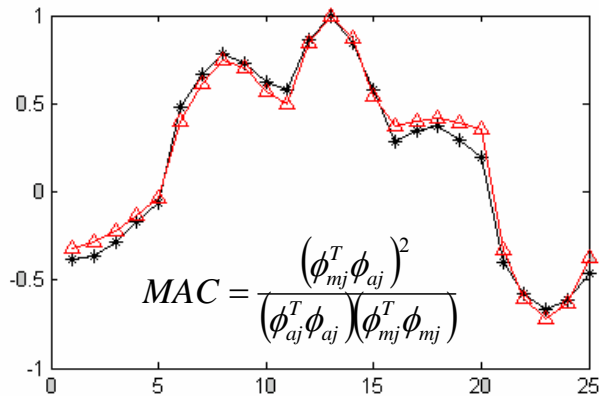


Natural Frequency Trend

Part 2-Model updating – damage localization



FE Model

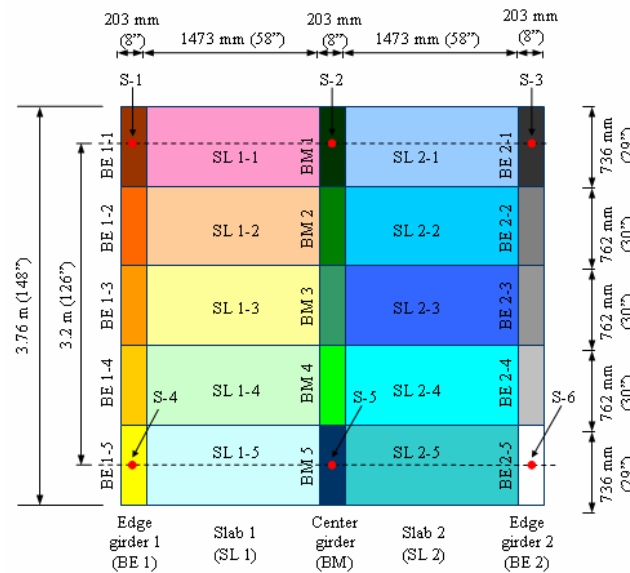


Match between Experiment and FEM Modal Amplitudes for 1st longitudinal Bending

FUNDAMENTAL SYSTEM ID EQUATION

$$Z = F\alpha \quad k_j^* = k_j(1 + \alpha_j)$$

- k_j^* = unknown stiffness of the j th member of structure for which M eigenvalues are known
- k_j = known stiffness of the j th member of FE model for which M eigenvalues are known
- α = $N \times 1$ matrix with fractional changes in stiffness between FE model and structure
- Z = $n \times 1$ matrix containing fractional changes in eigenvalues between two systems
- F = $n \times NE$ stiffness sensitivity matrix relating fractional changes in stiffness to eigenvalues



Parameters used:

- 5 Slab 1 stiffness
- 5 Slab 2 stiffness
- 5 Edge Beam 1 stiffness
- 5 Edge Beam 2 stiffness
- 5 Mid Beam 1 stiffness
- 6 spring stiffness

TOTAL of **31 parameters**

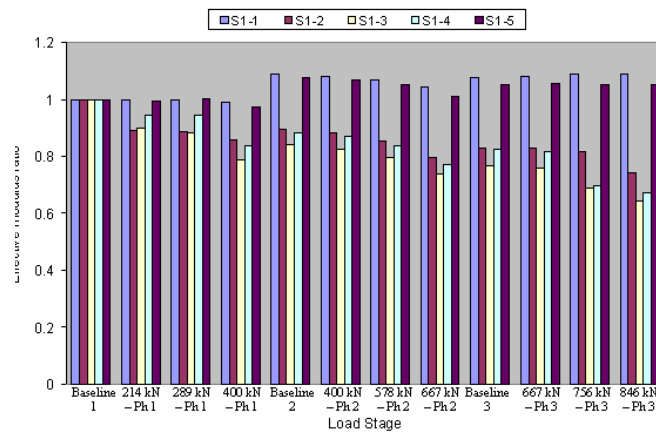
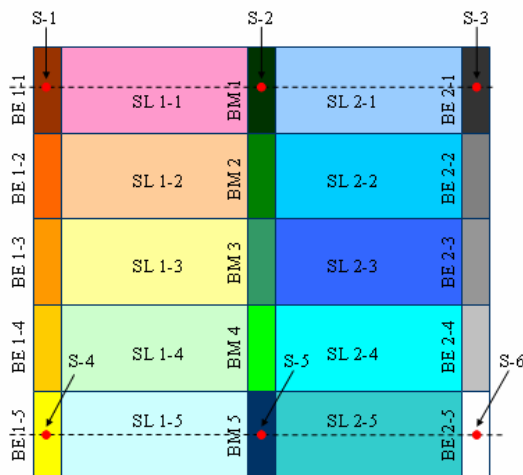
Targets for system ID → Measured **Frequencies** and **Modal Amplitudes** (n numbers)

$$Z = [n \times 1]$$

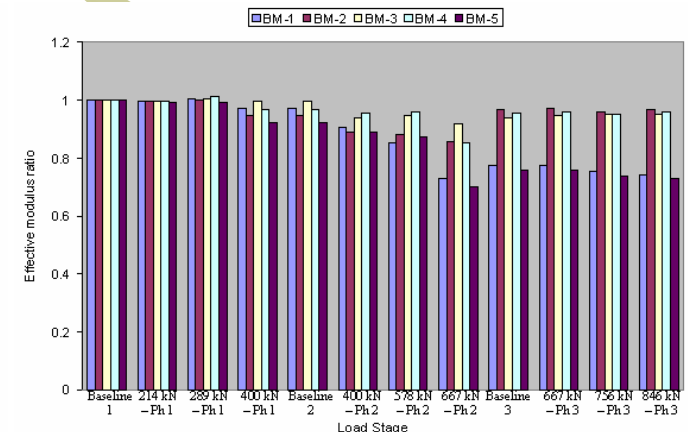
$$F = [n \times 31] \quad \longrightarrow \quad \alpha = [31 \times 1]$$

Part 2-Damage severity estimation

$$(Parameter\ Ratio)_i = \frac{Updated\ Parameter\ Value\ at\ Stage\ i\ of\ Test}{Updated\ Parameter\ Value\ at\ Baseline\ 1}$$

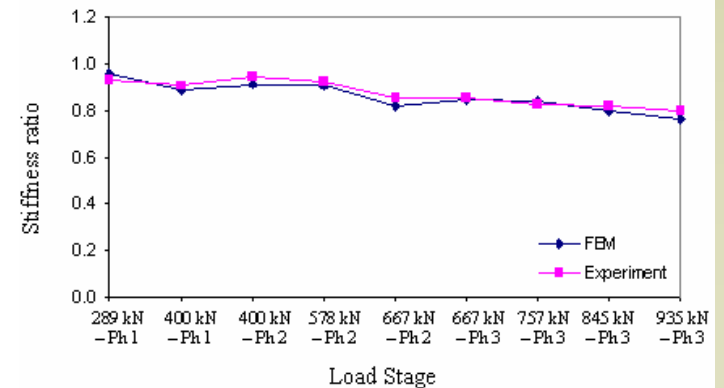


Changes in slab stiffness



Changes in girder stiffness

$$(Stiffness\ Ratio)_i = \frac{\left(\frac{Load, P_i\ at\ Stage\ i}{Deflection\ corresponding\ to\ Load, P_i\ at\ Stage\ i} \right)}{\left(\frac{Load, P_B\ at\ Baseline\ Load\ Stage}{Deflection\ corresponding\ to\ Load, P_B\ at\ Baseline\ Load\ Stage} \right)}$$



Comparison of stiffness ratios in Slab

Conclusions

- ◆ **An energy-based analytical model is proposed to predict the debonding failure in FRP-strengthened RC beams;**
- ◆ **Design considerations incorporating different failure modes are made in using FRP composites to retrofit/strengthen existing RC beams;**
- ◆ **NDE is needed to quantitatively monitor over time the appearance/progression of damage in composite strengthened structure**
- ◆ **A combination of Global and Local NDE techniques are required; Modal Testing and IR Thermography have shown promise to be implemented for field applications**



Questions / Comments ?